# "Seismic Analysis and Design of a four storey RC Building"

### **A PROJECT**

Submitted in partial fulfillment of the requirements for the award of the degree of

### **BACHELOR OF TECHNOLOGY**

IN

#### **CIVIL ENGINEERING**

Under the supervision of *Prof. Poonam Dhiman* 

By

Dilpuneet Singh (121614)

to



#### JAYPEE UNIVERSITY OF INFORMATION TECHNOLOGY

#### WAKNAGHAT SOLAN – 173 234

#### HIMACHAL PRADESH INDIA

June, 2016

# CERTIFICATE

This is to certify that the work which is being presented in the project title "Seismic Analysis and Design of RC building" in partial fulfilment of the requirements for the award of the degree of Bachelor of technology and submitted to Civil Engineering Department, Jaypee University of Information Technology, Waknaghat is an authentic record of work carried out by Dilpuneet Singh (121614) during a period from July 2015 to December 2015 under the supervision of Prof. Poonam Dhiman Assistant Professor, Civil Engineering Department, Jaypee University of Information Technology, Waknaghat.

The above statement made is correct to the best of my knowledge.

Date: -

Prof. Dr. Ashok Kumar Gupta Professor & Head of Department Civil Engineering Department JUIT Waknaghat Mrs. Poonam Dhiman Assistant Professor Civil Engineering Department JUIT Waknaghat

# ACKNOWLEDGEMENT

I would like to express my profound sense of deepest gratitude to my guide and motivator **Prof. Poonam Dhiman**, Assistant Professor (Grade-II), Civil Engineering Department, Jaypee University of Information Technology, Solan H.P. for her valuable guidance.

I wish to convey my sincere gratitude to all the faculty members of Civil Engineering Department who have enlightened me during my studies. The facilities and co-operation received from the technical staff of Civil Engineering Department is thankfully acknowledged.

Dilpuneet Singh 121614

# ABSTRACT

In this project first static and dynamic analysis of four storey reinforced concrete building was carried out. The calculations were carried out first manually for both the methods then same calculations were done using staad pro. The results of analysis in terms of member forces were compared and it was found that response spectrum dynamic analysis is more economical as compared to static. Design of beams and columns was carried out by preparing spread sheets for doubly reinforced beams and columns under biaxial bending. The seismic analysis and design is done which will ensure safe building during earthquake.

# CONTENTS

Chapter 1 Introduction1
Chapter 2 Methods of seismic analysis
2.1 Equivalent Static Lateral Force Method
2.2 Response Spectrum Method
2.3 Time History Method
Chapter 3 Design example problem5
3.1 Introduction
3.2 Equivalent Static Lateral Force Method
<ul> <li>3.2.1 Step1: Calculation of Lumped Masses to various floor levels</li> <li>3.2.2 Step 2: Determination of Fundamental Natural Period</li> <li>3.2.3 Step 3: Determination of Design Base Shear</li> <li>3.2.4 Step 4: Vertical Distribution of Base Shear</li> </ul>
3.3 Response Spectrum Method
A:Frame without considering the stiffness of infills
<ul> <li>3.3.1 Step 1: Determination of Eigenvalues and Eigenvectors</li> <li>3.3.2 Step 2: Determination of Modal Participation Factors</li> <li>3.3.3 Step 3: Determination of Modal Mass</li> <li>3.3.4 Step 4: Determination of lateral force at each floor in each mode</li> <li>3.3.5 Step 5: Determination of storey shear forces in each mode</li> <li>3.3.6 Step 6: Determination of storey shear force due to all modes</li> <li>3.3.7 Step 7: Determination of lateral forces at each storey</li> </ul>
B:Frame considering the stiffness of infills
Chapter 4 Seismic analysis by Staad Pro27
4.1 About Staad Pro
4.2 Static analysis 4.2.1 Source code 4.2.2 Results

4 2 2 1 Beam member forces
4.2.2.1 Dealin member forces
4.2.2.2 Colum member forces
4.3 Response spectrum analysis
4.3.1 Source code
4.3.2 Results
4.3.2.1 Beam member forces
4.3.2.2 Column member forces
Chapter 5 Design of beams and columns by spread sheet
5.1 Design of beams
5.2 Spread sheet of design of doubly reinforced beam
5.3 Design of beams of building
5.3 1 Rooms in transverse direction
5.3.1 Deanis in transverse unrection
5.5.2 Beams in longitudinal direction
5.4 Design of columns
5.5 Spread sheet of design of short columns under axial compression with biaxial bending
5.6 Design of columns of building
Conclusions

# Chapter 1 Introduction

### 1.Introduction

Earthquake-resistant structures are structures designed to withstand earthquakes. While no structure can be entirely immune to damage from earthquakes, the goal of earthquake-resistant construction is to erect structures that fare better during seismic activity than their conventional counterparts.

According to building codes, earthquake-resistant structures are intended to withstand the largest earthquake of a certain probability that is likely to occur at their location. This means the loss of life should be minimized by preventing collapse of the buildings for rare earthquakes while the loss of functionality should be limited for more frequent one.

Currently, there are several design philosophies in earthquake engineering, making use of experimental results, computer simulations and observations from past earthquakes to offer the required performance for the seismic threat at the site of interest. These range from appropriately sizing the structure to be strong and ductile enough to survive the shaking with an acceptable damage, to equipping it with base isolation or using structural vibration control technologies to minimize any forces and deformations. While the former is the method typically applied in most earthquake-resistant structures, important facilities, landmarks and cultural heritage buildings use the more advanced (and expensive) techniques of isolation or control to survive strong shaking with minimal damage. Examples of such applications are the Cathedral of Our Lady of the Angels and the Acropolis Museum.

In this project we will do seismic analysis of the four storey RC building with equivalent lateral force method and response spectrum method. In response spectrum we will first analysis without considering the infill wall and then with considering infill wall. We will calculate forces with both methods and then will compare the forces from the two methods. Our building is a plane frame model with four storeys. We will then do seismic analysis by staad pro and calculate results by both manually and staad pro. Then we will do design of beams and columns of the building.

# Chapter 2 Methods of Seismic Analysis

# 2. <u>Methods of seismic analysis</u>

Seismic analysis is a subset of structural analysis and is the calculation of the response of a building (or nonbuilding) structure to earthquakes. It is part of the process of structural design, earthquake engineering or structural assessment and retrofit in regions where earthquakes are prevalent.

#### 2.1 Equivalent Static Lateral Force Method

The concept employed in equivalent static lateral force procedures is to place static loads on a structure with magnitudes and direction that closely approximate the effects of dynamic loading caused by earthquakes. Concentrated lateral forces due to dynamic loading tend to occur at floor and ceiling/roof levels in buildings, where concentration of mass is the highest. Furthermore, concentrated lateral forces tend to be larger at higher elevations in a structure. Thus, the greatest lateral displacements and the largest lateral forces often occur at the top level of a structure (particularly for tall buildings).

#### 2.2 Response Spectrum Method

With the advent of personal computers and improved structural analysis techniques, the use of more precise methods increased. One of the most popular was response spectrum analysis. The method requires the determination of a response spectrum from measured seismic activity. This data was then reduced into a spectrum of seismic action versus natural frequency. The seismic action could be displacement, velocity, or acceleration, although the typical value used was acceleration. Detailed information from the structural model was coupled with the corresponding spectral values for each specific mode of vibration. The independent results were then combined using an appropriate technique to determine the response of the overall structure.

#### 2.3 Time History Method

Time-History analysis is a step-by-step procedure where the loading and the response history are evaluated at successive time increments,  $\Delta t$  – steps. During each step the response is evaluated from the initial conditions existing at the beginning of the step (displacements and velocities) and the loading history in the interval. With this method the non-linear behaviour may be easily considered by changing the structural properties (e.g. stiffness, k) from one step to the next. Therefore this method is one of the most effective for the solution of non-linear response, among the many methods available. Nevertheless, in the present text, a linear time history analysis is adopted i.e. the structural properties are assumed to remain constant during the entire loading history and further it is assumed that the structure behaves linearly.

# Chapter 3 Design Example Problem

# 3. <u>Design Example Problem</u>

#### **3.1 Introduction**

A four storey RC building has been analysed by the equivalent static method, response spectrum method and time history method as per IS 1893 (Part 1): 2002. The example illustrates step by step procedure for determination of forces. One of the plane frame in transverse direction has been considered for the purpose of illustration by assuming that the building is symmetric in elevation and planned as shown in figure 3.1. The preliminary building data required for analysis are assumed in table 3.1.The building is evaluated by three methods i.e. Equivalent static lateral force method, response spectrum method and time history method.



Fig. 3.1 Plane frame structure model

#### Table 3.1 Assumed preliminary data required for analysis of frame

1.	Type of structure	Multi storey rigid joined plane frame	
2.	Seismic zone	V (table 2, IS 1893 (Part1): 2002)	
3.	Number of storeys	Four (G+3)	
4.	Floor height	3.5m	
5.	Infill wall	300mm thick including plaster in	
		longitudinal and 200mm in transverse	
		direction	
6.	Imposed load	3.5kN/m <sup>2</sup>	
7.	Materials	Concrete (M 20) and Reinforcement	
		(Fe415)	
8.	Size of columns	300mm × 500mm	
9.	Size of beam	300mm × 450mm in longitudinal and	
		300mm × 400mm in transverse direction	
10.	Depth of slab	150mm thick	
11	Specific weight of RCC	$25kN/m^2$	
11.			
12.	Specific weight of infill	20kN/m <sup>2</sup>	
13.	Type of soil	Rock	
14.	Response spectra	As per IS 1893 (Part 1): 2002	
15.	Time History	Compatible to IS 1893 (Part 1): 2002	
		spectra at rocky site for 5% damping	

#### **3.2 Equivalent Static Lateral Force Method**

A step by step procedure for analysis of the frame by equivalent static lateral force method is as follow:

#### **3.2.1 Step1: Calculation of Lumped Masses to various floor levels**

The earthquake forces shall be calculated for the full dead load plus the percentage of imposed load as given in Table 8 of IS 1893 (Part 1): 2002. The imposed load on roof is assumed to be zero. The lumped masses of each floor are worked out as follows:

#### Roof

Mass of infill + Mass of columns + Mass of beams in longitudinal and transverse direction of that floor + Mass of slab + Imposed load of that floor if permissible.

 $= \{((0.3 \times 10 \times (3.5/2) + 0.2 \times 15 \times (3.5/2)) \ 20\} + \{(0.3 \times 10 \times 0.45 + 0.3 \times 15 \times 0.4) \ 25\} + \{0.15 \times 5 \times 10 \times 25\} + \{(0.3 \times 0.5 \times (3.5/2) \times 3) \times 25\} + 0^*$ 

= 495.9375kN (weight) = 50.57ton (mass)

#### 3<sup>rd</sup>, 2<sup>nd</sup>, 1<sup>st</sup> Floors

 $= \{((0.3 \times 10 \times 3.5) + (0.2 \times 15 \times 3.5)) \ 20\} + \{(0.3 \times 10 \times 0.45 + 0.3 \times 15 \times 0.4) \ 25\} + \{0.15 \times 5 \times 10 \times 25\} + \{0.3 \times 0.5 \times 3.5 \times 3 \times 25\} + \{5 \times 10 \times 3.5 \times 0.5^{**}\}$ 

= 813.125kN (weight) = 82.91ton (mass)

\*imposed load on roof not considered. \*\*50% of imposed load, if imposed load is greater than 3kN/m<sup>2</sup>.

#### Seismic weight of building

= Seismic weight of all floors =  $M_1 + M_2 + M_3 + M_4$ = 82.91 + 82.91 + 82.91 + 50.57 = 299.3 ton

#### 3.2.2 Step 2: Determination of Fundamental Natural Period

The approximate fundamental natural period of a vibration  $(T_a)$ , in seconds, of a moment resisting frame building without brick infill panels may be estimated by the empirical expression

 $T_a = 0.075 \times h^{0.75} = 0.075 \times 14^{0.75} = 0.5423 sec$ 

where h is the height of the building, in meters.

#### 3.2.3 Step 3: Determination of Design Base Shear

Design seismic base shear,  $V_B = A_h \times W$ 

 $A_h = (Z \times I \times S_a) / (2 \times R \times g) = (0.36 \times 1 \times 1.842) / (2 \times 5) = 0.066312$ 

For  $T_a = 0.5423$ 

 $S_a/g = 1/T_a = 1.842$ , for rock site from figure 2 of IS 1893 (Part 1): 2002

Design seismic base shear,  $V_B = 0.066312 \times (299.3 \times 9.81) = 194.7$ kN

#### 3.2.4 Step 4: Vertical Distribution of Base Shear

The design base shear  $(V_B)$  computed shall be distributed along the height of the building as per the expression,

$$Q_i = V_B \times W_i \times h_i^2 / \sum_{i=1}^n W_i h_i^2$$
(3.1)

where,

 $Q_i$  = Design lateral forces at floor i,

 $W_i$  = Seismic weights of the floor i,

 $h_i$  = Height of the floor i, measured from base, and

#### n = Number of stories

Using the equation 3.1, base shear is distributed as follows:

$$Q_{1} = V_{B} (W_{1}h_{1}^{2} / W_{1}h_{1}^{2} + W_{2}h_{2}^{2} + W_{3}h_{3}^{2} + W_{4}h_{4}^{2})$$
  
= 194.7 (813.125 × 3.5<sup>2</sup> / 813.125 × 3.5<sup>2</sup> + 813.125 × 7<sup>2</sup> + 813.125 × 10.5<sup>2</sup> + 495.9375 × 14<sup>2</sup>)  
= 8.19kN

Similarly,

 $Q_2 = 0.1684 \times 194.7 = 32.78kN$   $Q_3 = 0.3788 \times 194.7 = 73.75kN$  $Q_4 = 0.4107 \times 194.7 = 79.96kN$ 



Fig. 3.2 Lateral force distribution at various floor level

# 3.3 Response Spectrum Method

#### A:Frame without considering the stiffness of infills

A step by step procedure for analysis of the frame by response spectrum method is as follows:

#### **3.3.1 Step 1: Determination of Eigenvalues and Eigenvectors**

Mass matrix, M and stiffness matrix, K of the plane frame lumped mass model are,

$$\mathbf{M} = \begin{bmatrix} M_1 & 0 & 0 & 0 \\ 0 & M_2 & 0 & 0 \\ 0 & 0 & M_3 & 0 \\ 0 & 0 & 0 & M_4 \end{bmatrix} = \begin{bmatrix} 82.91 & 0 & 0 & 0 \\ 0 & 82.91 & 0 & 0 \\ 0 & 0 & 82.91 & 0 \\ 0 & 0 & 0 & 50.57 \end{bmatrix}$$
ton

Column stiffness of storey,

$$k = 12EI / L^{3} = (12 \times 22360 \times 10^{3} \times (0.3 \times 0.5^{3} / 12)) / 3.5^{3} = 19556.85 kN/m$$

Total lateral stiffness of each storey,

 $k_1 = k_2 = k_3 = k_4 = 3 \times 19556.85 = 58670.55 \text{kN/m}$ 

Stiffness of lumped mass modified structure

$$\mathbf{K} = \begin{bmatrix} k_1 + k_2 & -k_2 & 0 & 0 \\ -k_2 & k_2 + k_3 & -k_3 & 0 \\ 0 & -k_3 & k_3 + k_4 & -k_4 \\ 0 & 0 & -k_4 & k_4 \end{bmatrix}$$
$$= \begin{bmatrix} 117341.1 & -58670.55 & 0 & 0 \\ -58670.55 & 117341.1 & -58670.55 & 0 \\ 0 & -58670.55 & 117341.1 & -58670.55 \\ 0 & 0 & -58670.55 & 58670.55 \end{bmatrix} \mathbf{kN/m}$$

For the above stiffness and mass matrix, eigenvalues and eigenvectors are worked out as follows:

$$\begin{vmatrix} K - \omega^2 m \end{vmatrix} = \begin{vmatrix} 2k - \omega^2 m & -k_2 & 0 & 0 \\ -k_2 & 2k - \omega^2 m & -k_3 & 0 \\ 0 & -k_3 & 2k - \omega^2 m & -k_4 \\ 0 & 0 & -k_4 & k - \omega^2 m \end{vmatrix} = 0$$
  
Taking k/m =  $\omega_n^2$   
Therefore,  
 $(\omega_n^2)^4 - 8.3(\omega_n^2)^3 (\omega^2) + 10.75(\omega_n^2)^2 (\omega^2)^2 - 4.45(\omega_n^2)(\omega^2)^3 + 0.575(\omega^2)^4 = 0$   
By solving the above equation, natural frequencies (eigenvalues) of various modes are

#### Eigenvalues

$$[\omega^{2}] = \begin{bmatrix} 81 \\ 657 \\ 1475 \\ 2065 \end{bmatrix}$$

$$\omega_1^2 = 81, \omega_2^2 = 657, \omega_3^2 = 1475, \omega_4^2 = 2065$$

The quantity of  $\omega_i^2$ , is called the i<sup>th</sup> eigenvalue of the matrix  $\left[-M\omega_i^2 + \kappa\right]\Phi_i$ . Each natural frequency  $(\omega_i)$  of the system has a corresponding eigenvector (mode shape), which is denoted by  $\Phi_i$ . The mode shape corresponding to each natural frequency is determined from the equations

$$\begin{bmatrix} -M\omega_1^2 + K \end{bmatrix} \Phi_1 = 0$$
$$\begin{bmatrix} -M\omega_2^2 + K \end{bmatrix} \Phi_2 = 0$$
$$\begin{bmatrix} -M\omega_3^2 + K \end{bmatrix} \Phi_3 = 0$$
$$\begin{bmatrix} -M\omega_4^2 + K \end{bmatrix} \Phi_4 = 0$$

Solving the above equation, modal vector (eigenvectors), mode shapes and natural periods under different modes are

# Eigenvectors {Φ}

	-0.0328	0.0795	0.0808	-0.0397
(A) - (A A A A )	-0.0608	0.0644	-0.0540	0.0690
$\{\Phi\} = \{\Phi_1 \ \Phi_2 \ \Phi_3 \ \Phi_4\} =$	-0.0798	-0.0273	-0.0448	-0.0799
	-0.0872	-0.0865	0.0839	0.0696

# **Mode Shapes**



Fig. 3.3 Modal Shapes

Natural time period

$$T = \begin{bmatrix} 0.6977 & 0 & 0 & 0 \\ 0 & 0.2450 & 0 & 0 \\ 0 & 0 & 0.1636 & 0 \\ 0 & 0 & 0 & 0.1383 \end{bmatrix} s$$

# **3.3.2 Step 2: Determination of Modal Participation Factors**

The modal participation factor  $(p_k)$  of mode k is,

$$p_{k} = \frac{\sum_{i=1}^{n} W_{i} \phi_{ik}}{\sum_{i=1}^{n} W_{i} (\phi_{ik})^{2}}$$

$$p_{1} = \frac{\sum_{i=1}^{4} W_{i} \phi_{i1}}{\sum_{i=1}^{4} W_{i} (\phi_{i1})^{2}} = \frac{(W_{1} \phi_{11} + W_{2} \phi_{21} + W_{3} \phi_{31} + W_{4} \phi_{41})}{(W_{1} (\phi_{11})^{2} + W_{2} (\phi_{21})^{2} + W_{3} (\phi_{31})^{2} + W_{4} (\phi_{41})^{2})} = -14.45$$

$$p_{2} = \frac{\sum_{i=1}^{4} W_{i} \phi_{i2}}{\sum_{i=1}^{4} W_{i} (\phi_{i2})^{2}} = \frac{(W_{1} \phi_{12} + W_{2} \phi_{22} + W_{3} \phi_{32} + W_{4} \phi_{42})}{(W_{1} (\phi_{12})^{2} + W_{2} (\phi_{22})^{2} + W_{3} (\phi_{32})^{2} + W_{4} (\phi_{42})^{2})} = 4.06$$

Similarly,

 $p_3 = 2.1, p_4 = -0.52$ 

# **3.3.3 Step 3: Determination of Modal Mass**

The modal mass  $(M_k)$  of mode k is given by,

$$M_{k} = \frac{\left[\sum_{i=1}^{n} W_{i} \phi_{ik}\right]^{2}}{g\left[\sum_{i=1}^{n} W_{i} (\phi_{ik})^{2}\right]}$$

where,

g = Acceleration due to gravity,  $\Phi_{ik} = Mode shape coefficient at floor i in mode k, and$   $W_i = Seismic weight of floor i,$ 

$$\mathbf{M}_{1} = \frac{\left[\sum_{i=1}^{4} W_{i} \phi_{i1}\right]^{2}}{g\left[\sum_{i=1}^{4} W_{i} (\phi_{i1})^{2}\right]}$$

 $M_1$ 

$$= \frac{\left[9.81(82.91(-0.0328) + 82.91(-0.0608) + 82.91(-0.0798) + 50.57(-0.0872))\right]^2}{9.81\left[9.81\left(82.91(-0.0328)^2 + 82.91(-0.0608)^2 + 82.91(-0.0798)^2 + 50.57(-0.0872)^2\right)\right]} = 269.85$$

$$M_{2} = \frac{\left[\sum_{i=1}^{4} W_{i} \phi_{i2}\right]^{2}}{g\left[\sum_{i=1}^{4} W_{i} (\phi_{i2})^{2}\right]}, \text{ similarly, } M_{2} = 21.42, M_{3} = 5.78, M_{4} = 0.34$$

#### Modal contributions of various modes

For mode 1,  $\frac{M_1}{M} = \frac{269.85}{299.3} = 0.90 = 90\%$ 

For mode 2, 
$$\frac{M_2}{M} = \frac{21.42}{299.3} = 0.0715 = 7.15\%$$

For mode 3, 
$$\frac{M_3}{M} = \frac{5.78}{299.3} = 0.0193 = 1.93\%$$

For mode 4,  $\frac{M_4}{M} = \frac{0.34}{299.3} = 0.0011 = 0.11\%$ 

#### 3.3.4 Step 4: Determination of lateral force at each floor in each mode

The design lateral force  $(Q_{ik})$  at floor i in mode k is given by,

 $Q_{ik} = A_k \Phi_{ik} P_k W_i$ 

where,

 $A_k$  = Design horizontal acceleration spectrum value as per clause 6.4.2 of IS 1893 (Part 1): 2002 using the natural period of vibration ( $T_k$ ) of mode k,

The design horizontal seismic coefficient Ah for various modes are,

$$A_{hk} = \frac{Z}{2} \frac{I}{R} \frac{S_{ak}}{g}$$

$$A_{h1} = \frac{Z}{2} \frac{I}{R} \frac{S_{a1}}{g} = \frac{0.36}{2} \frac{1}{5} \quad 1.433 = 0.0515$$

$$A_{h2} = \frac{Z}{2} \frac{I}{R} \frac{S_{a2}}{g} = \frac{0.36}{2} \frac{1}{5} \quad 2.5 = 0.09$$

Similarly  $A_{h3} = 0.09$ ,  $A_{h4} = 0.09$ 

The average response acceleration coefficient for rock sites as per IS 1893 (Part 1): 2002 is calculated as follows:

For rocky, or hard soil sites

$$\frac{S_a}{g} = \begin{cases} 1+1.5T; & 0.00 \le T \le 0.10\\ 2.5; & 0.10 \le T \le 0.40\\ 1.00/T; & 0.40 \le T \le 4.0 \end{cases}$$

For 
$$T_1 = 0.6978 \implies \frac{S_{a1}}{g} = 1.433$$
  
For  $T_2 = 0.2450 \implies \frac{S_{a2}}{g} = 2.5$   
For  $T_3 = 0.1636 \implies \frac{S_{a3}}{g} = 2.5$ 

For 
$$T_4 = 0.1382 \implies \frac{S_{a4}}{g} = 2.5$$

#### Design lateral force in each mode

$$\begin{aligned} Q_{i1} &= (A_1 P_1 \Phi_{i1} W_i) \\ & [Q_{i1}] = \begin{bmatrix} (A_{h1} P_1 \Phi_{11} W_1) \\ (A_{h1} P_1 \Phi_{21} W_2) \\ (A_{h1} P_1 \Phi_{31} W_3) \\ (A_{h1} P_1 \Phi_{41} W_4) \end{bmatrix} = \\ & \begin{bmatrix} ((0.0515) (-14.45) (-0.0328) (82.91 \times 9.81)) \\ ((0.0515) (-14.45) (-0.0608) (82.91 \times 9.81)) \\ ((0.0515) (-14.45) (-0.0798) (82.91 \times 9.81)) \\ ((0.0515) (-14.45) (-0.0872) (50.57 \times 9.81)) \end{bmatrix} \end{aligned}$$

$$= \begin{bmatrix} (19.852) \\ (36.8) \\ (48.3) \\ (32.192) \end{bmatrix} kN$$
  
Similarly,  $[Q_{i2}] = \begin{bmatrix} 23.62 \\ 19.13 \\ -8.11 \\ -15.68 \end{bmatrix}, [Q_{i3}] = \begin{bmatrix} 12.42 \\ -8.30 \\ -6.88 \\ 7.86 \end{bmatrix}, [Q_{i4}] = \begin{bmatrix} 1.511 \\ -2.62 \\ 3.04 \\ -1.615 \end{bmatrix}$ 

#### 3.3.5 Step 5: Determination of storey shear forces in each mode

The peak shear force is given by,

$$\mathbf{V}_{ik} = \sum_{j=i+1}^{n} Q_{ik}$$

The storey shear forces for the first mode is,

$$\mathbf{V}_{i1} = \sum_{j=i+1}^{n} Q_{i1} = \begin{bmatrix} V_{11} \\ V_{21} \\ V_{31} \\ V_{41} \end{bmatrix} = \begin{bmatrix} (Q_{11} + Q_{21} + Q_{31} + Q_{41}) \\ (Q_{21} + Q_{31} + Q_{41}) \\ (Q_{31} + Q_{41}) \\ (Q_{41}) \end{bmatrix} = \begin{bmatrix} 137.144 \\ 117.292 \\ 80.492 \\ 32.192 \end{bmatrix} \mathbf{kN}$$

Similarly,

$$v_{i2} = \begin{bmatrix} 18.96 \\ -4.66 \\ -23.79 \\ -15.68 \end{bmatrix}, v_{i3} = \begin{bmatrix} 5.1 \\ -7.32 \\ 0.98 \\ 7.86 \end{bmatrix}, v_{i4} = \begin{bmatrix} 0.316 \\ -1.195 \\ 1.425 \\ -1.615 \end{bmatrix}$$

#### 3.3.6 Step 6: Determination of storey shear force due to all modes

The peak storey force  $(V_1)$  in storey i due to all modes considered is obtained by combining those due to each mode in accordance with modal combination i.e. SRSS (Square Root of Sum of Squares) or CQC (Complete Quadratic Combination) methods.

#### Square root of sum of squares (SRSS)

If the building does not have closely spaced modes, the peak response quality ( $\lambda$ ) due to all modes considered shall be obtained as,

$$\lambda = \sqrt{\sum_{k=1}^{r} (\lambda_k)^2} ,$$

where,

 $\lambda_k$  = Absolute value of quantity in mode 'k', and r is the numbers of modes being considered Using the above method, the storey shears are,

$$V_{1} = [(V_{11})^{2} + (V_{12})^{2} + (V_{13})^{2} + (V_{14})^{2}]^{1/2}$$
  

$$= [(137.144)^{2} + (18.96)^{2} + (5.1)^{2} + (0.316)^{2}]^{1/2} = 138.54 \text{kN}$$

$$V_{2} = [(V_{21})^{2} + (V_{22})^{2} + (V_{23})^{2} + (V_{24})^{2}]^{1/2}$$
  

$$= [(117.292)^{2} + (-4.66)^{2} + (-7.32)^{2} + (-1.195)^{2}]^{1/2} = 117.61 \text{kN}$$

$$V_{3} = [(V_{31})^{2} + (V_{32})^{2} + (V_{33})^{2} + (V_{34})^{2}]^{1/2}$$
  

$$= [(80.492)^{2} + (-23.79)^{2} + (0.98)^{2} + (1.425)^{2}]^{1/2} = 83.95 \text{kN}$$

$$V_{4} = [(V_{41})^{2} + (V_{42})^{2} + (V_{43})^{2} + (V_{44})^{2}]^{1/2}$$
  

$$= [(32.192)^{2} + (-15.68)^{2} + (7.86)^{2} + (-1.615)^{2}]^{1/2} = 36.69 \text{kN}$$

#### 3.3.7 Step 7: Determination of lateral forces at each storey

The design lateral forces  $F_{roof}$  and  $F_i$ , at roof and the  $i^{th}$  floor, are calculated as,

$$F_{roof} = V_{roof} \quad and \ F_i = V_i \text{ - } V_{i+1}$$

#### Square root of sum of squares (SRSS)

 $F_{roof} = F_4 = V_4 = 36.69 kN$ 

 $F_{floor3} = F_3 = V_3 - V_4 = 83.95 - 36.69 = 47.26 \text{kN}$ 

 $F_{floor2} = F_2 = V_2 - V_3 = 117.61 - 83.95 = 33.66 \text{kN}$ 

 $F_{floor1} = F_1 = V_1 - V_2 = 138.54 - 117.61 = 20.93 kN$ 

#### **B:Frame considering the stiffness of infills**

The frame considered in previous section is again analysed by considering the stiffness of infill walls. The infill is modelled as equivalent diagonal strut.

The mass matrix [M] for the lumped plane frame model is

$$\mathbf{M} = \begin{bmatrix} M_1 & 0 & 0 & 0 \\ 0 & M_2 & 0 & 0 \\ 0 & 0 & M_3 & 0 \\ 0 & 0 & 0 & M_4 \end{bmatrix} = \begin{bmatrix} 82.91 & 0 & 0 & 0 \\ 0 & 82.91 & 0 & 0 \\ 0 & 0 & 82.91 & 0 \\ 0 & 0 & 0 & 50.57 \end{bmatrix}$$
ton

Column stiffness of storey,

$$k = 12EI / L^3 = (12 \times 22360 \times 10^3 \times (0.3 \times 0.5^3 / 12)) / 3.5^3 = 19556.85 kN/m$$

Stiffness of infill is determined by modelling the infill as an equivalent diagonal strut in which

Width of strut, W=  $\frac{1}{2}\sqrt{\alpha_h^2 + \alpha_l^2}$  $\alpha_{\rm h} = \frac{\Pi}{2} \left[ \frac{E_f I_c h}{2E_f t \sin 2\theta} \right]^{\frac{1}{4}}, \alpha_{\rm l} = \Pi \left[ \frac{E_f I_b l}{E_f t \sin 2\theta} \right]^{\frac{1}{4}}, \theta = \tan^{-1} \frac{h}{l}$ Where,  $E_f$  = Elastic modulus of frame material = 22360N/m<sup>2</sup>  $E_m$ = Elastic modulus of masonry wall = 13800N/m<sup>2</sup> t = Thickness of infill wall = 300mm h = Height of infill wall = 3.5m I = Length of infill wall = 5m $I_c$  = Moment of inertia of columns =  $\frac{.3 \times .5^3}{12}$  = 0.003125 m<sup>4</sup>  $I_b$  = Moment of inertia of beams =  $\frac{.3 \times .45^3}{12}$  = 0.002278 m<sup>4</sup>  $\left| \alpha_{h} = \frac{\Pi}{2} \left[ \frac{22360 \times 0.003125 \times 3.5}{2 \times 13800 \times 0.3 \times \sin 2(35)} \right]^{\frac{1}{4}} = 0.69 \text{ m}$  $\alpha_{\rm I} = \Pi \left[ \frac{22360 \times 0.002278 \times 5}{13800 \times 0.3 \times \sin 2(35)} \right]^{\frac{1}{4}} = 1.66 \text{ m}$ W=  $\frac{1}{2}\sqrt{\alpha_h^2 + \alpha_l^2} = 0.8988 \text{ m}$ 

A = Cross sectional area of diagonal stiffness =  $W \times t = 0.8988 \times 0.3 = 0.2696 \text{m}^2$ 

 $l_{d} = \text{Diagonal length of strut} = \sqrt{h^{2} + l^{2}} = 6.103 \text{ m}$ Therefore, stiffness of infill is  $\frac{AE_{m}}{l_{d}}\cos^{2}\theta = \frac{0.2696 \times 13800 \times 10^{6}}{6.103} 0.819^{2} = 408905.929 \times 10^{3} \text{ N/m}$ 

For the frame with two bays there are two struts participating in one direction, total lateral stiffness of each storey

 $k_1 = k_2 = k_3 = k_4 = 3 \times 19556.85 + 2 \times 408905929 = 817870.5286 kN/m$ 

Stiffness matrix [K] of lumped mass model is,

$$\mathbf{K} = \begin{bmatrix} k_1 + k_2 & -k_2 & 0 & 0 \\ -k_2 & k_2 + k_3 & -k_3 & 0 \\ 0 & -k_3 & k_3 + k_4 & -k_4 \\ 0 & 0 & -k_4 & k_4 \end{bmatrix} = \begin{bmatrix} 1.6357 & -0.8178 & 0 & 0 \\ -0.8178 & 1.6357 & -0.8178 & 0 \\ 0 & -0.8178 & 1.6357 & -0.8178 \\ 0 & 0 & -0.8178 & 0.8178 \end{bmatrix} \times 10^6 \text{kN/m}$$

For the above stiffness and mass matrices, eigenvalues and eigenvectors are,

$$\begin{vmatrix} K - \omega^2 m \end{vmatrix} = \begin{vmatrix} 2k - \omega^2 m & -k_2 & 0 & 0 \\ -k_2 & 2k - \omega^2 m & -k_3 & 0 \\ 0 & -k_3 & 2k - \omega^2 m & -k_4 \\ 0 & 0 & -k_4 & k - \omega^2 m \end{vmatrix} = 0, \ k/m = \omega_n^2$$

Therefore, quadratic equation is,

$$(\omega_n^2)^4 - 8.3(\omega_n^2)^3 (\omega^2) + 10.75(\omega_n^2)^2 (\omega^2)^2 - 4.45(\omega_n^2)(\omega^2)^3 + 0.575(\omega^2)^4 = 0$$

# **Eigenvalues** 1442 $\begin{bmatrix} \omega^2 \end{bmatrix} = \begin{bmatrix} 1442 \\ 11698 \\ 26227 \\ 36719 \end{bmatrix}$ $\omega_1^2 = 1442, \omega_2^2 = 11698, \omega_3^2 = 26227, \omega_4^2 = 36719$ **Eigenvectors** {Φ} $\{\Phi\} = \{\Phi_1 \ \Phi_2 \ \Phi_3 \ \Phi_4 \} = \begin{bmatrix} -0.0328 & 0.0795 & 0.0808 & -0.0397 \\ -0.0608 & 0.0644 & -0.0540 & 0.0690 \\ -0.0798 & -0.0273 & -0.0448 & -0.0799 \\ -0.0872 & -0.0865 & 0.0839 & 0.0696 \end{bmatrix}$ -0.0397Natural frequency in various modes 0 37.975 0 0 $\begin{array}{ccccccc} 0 & 108.157 & 0 & 0 \\ 0 & 0 & 161.947 & 0 \\ 0 & 0 & 0 & 191.621 \end{array}$ rad/s [ω]=

Natural time period

$$T = \begin{bmatrix} 0.1655 & 0 & 0 & 0 \\ 0 & 0.0581 & 0 & 0 \\ 0 & 0 & 0.0388 & 0 \\ 0 & 0 & 0 & 0.0328 \end{bmatrix} s$$

**Modal Participation Factors** 

$$p_{k} = \frac{\sum_{i=1}^{n} W_{i} \phi_{ik}}{\sum_{i=1}^{n} W_{i} (\phi_{ik})^{2}}$$

$$\begin{split} & \left[ p_{1} = \frac{\sum_{i=1}^{4} W_{i} \phi_{1}}{\sum_{i=1}^{4} W_{i} (\phi_{1})^{2}} = \frac{(W_{1} \phi_{11} + W_{2} \phi_{21} + W_{3} \phi_{31} + W_{4} \phi_{41})}{(W_{1} (\phi_{11})^{2} + W_{2} (\phi_{21})^{2} + W_{3} (\phi_{31})^{2} + W_{4} (\phi_{41})^{2})} = -14.45 \\ & p_{2} = \frac{\sum_{i=1}^{4} W_{i} \phi_{2}}{\sum_{i=1}^{4} W_{i} (\phi_{2})^{2}} = \frac{(W_{1} \phi_{12} + W_{2} \phi_{22} + W_{3} \phi_{32} + W_{4} \phi_{42})}{(W_{1} (\phi_{12})^{2} + W_{2} (\phi_{22})^{2} + W_{3} (\phi_{32})^{2} + W_{4} (\phi_{42})^{2})} = 4.06 \\ & \text{Similarly,} \\ & p_{3} = 2.1, p_{4} = -0.52 \\ & \text{Modal Mass} \\ & M_{k} = \frac{\left[\sum_{i=1}^{n} W_{i} \phi_{k}\right]^{2}}{g\left[\sum_{i=1}^{n} W_{i} (\phi_{k})^{2}\right]} \\ & M_{i} = \frac{\left[\sum_{i=1}^{n} W_{i} \phi_{k}\right]^{2}}{g\left[\sum_{i=1}^{n} W_{i} (\phi_{k})^{2}\right]} \\ & M_{i} = \frac{\left[9.81(82.91(-0.0328) + 82.91(-0.0608) + 82.91(-0.0798) + 50.57(-0.0872))\right]^{2}}{9.81\left[9.81(82.91(-0.0328) + 82.91(-0.0608)^{2} + 82.91(-0.0798)^{2} + 50.57(-0.0872))\right]^{2}} \\ & = 269.85 \\ & M_{2} = \frac{\left[\sum_{i=1}^{4} W_{i} \phi_{2}\right]^{2}}{g\left[\sum_{i=1}^{4} W_{i} \phi_{2}\right]^{2}}, \text{ similarly, } M_{2} = 21.42, M_{3} = 5.78, M_{4} = 0.34 \end{aligned}$$

#### Modal contributions of various modes

For mode 1,  $\frac{M_1}{M} = \frac{269.85}{299.3} = 0.90 = 90\%$ For mode 2,  $\frac{M_2}{M} = \frac{21.42}{299.3} = 0.0715 = 7.15\%$ For mode 3,  $\frac{M_3}{M} = \frac{5.78}{299.3} = 0.0193 = 1.93\%$ 

For mode 4,  $\frac{M_4}{M} = \frac{0.34}{299.3} = 0.0011 = 0.11\%$ 

#### Design lateral force at each floor in each mode

The design lateral force  $(Q_{ik})$  at floor i in mode k is given by,

 $Q_{ik} \,{=}\, A_k \; \Phi_{ik} \; P_k \; W_i$ 

The design horizontal seismic coefficient Ah for various modes are,

$$A_{hk} = \frac{Z}{2} \frac{I}{R} \frac{S_{ak}}{g}$$

$$A_{h1} = \frac{Z}{2} \frac{I}{R} \frac{S_{a1}}{g} = \frac{0.36}{2} \frac{1}{5} 2.5 = 0.090$$

$$A_{h2} = \frac{Z}{2} \frac{I}{R} \frac{S_{a2}}{g} = \frac{0.36}{2} \frac{1}{5} 1.871 = 0.067$$

Similarly  $A_{h3} = 0.056$ ,  $A_{h4} = 0.053$ 

The average response acceleration coefficient for rock sites as per IS 1893 (Part 1): 2002 is calculated as follows:

For rocky, or hard soil sites

$$\frac{S_a}{g} = \begin{cases} 1+1.5T; & 0.00 \le T \le 0.10\\ 2.5; & 0.10 \le T \le 0.40\\ 1.00/T; & 0.40 \le T \le 4.0 \end{cases}$$

For 
$$T_1 = 0.1655 \implies \frac{S_{a1}}{g} = 2.5$$
  
For  $T_2 = 0.0581 \implies \frac{S_{a2}}{g} = 1+15T = 1.871$   
For  $T_3 = 0.0388 \implies \frac{S_{a3}}{g} = 1+15T = 1.582$   
For  $T_4 = 0.1382 \implies \frac{S_{a4}}{g} = 1+15T = 1.49$ 

# Design lateral force in each mode

$$Q_{11} = (A_{1} P_{1} \Phi_{11} W_{i})$$

$$[Q_{11}] = \begin{bmatrix} (A_{h1} P_{1} \Phi_{11} W_{1}) \\ (A_{h1} P_{1} \Phi_{21} W_{2}) \\ (A_{h1} P_{1} \Phi_{31} W_{3}) \\ (A_{h1} P_{1} \Phi_{41} W_{4}) \end{bmatrix} =$$

$$\begin{bmatrix} ((0.090) (-14.45) (-0.0328) (82.91 \times 9.81)) \\ ((0.090) (-14.45) (-0.0608) (82.91 \times 9.81)) \\ ((0.090) (-14.45) (-0.0798) (82.91 \times 9.81)) \\ ((0.090) (-14.45) (-0.0872) (50.57 \times 9.81)) \end{bmatrix}$$

$$= \begin{bmatrix} (34.69) \\ (64.31) \\ (84.40) \\ (56.25) \end{bmatrix} kN$$
Similarly,  $[Q_{12}] = \begin{bmatrix} 17.58 \\ 14.24 \\ -6.04 \\ -11.67 \end{bmatrix}, [Q_{13}] = \begin{bmatrix} 7.728 \\ -5.16 \\ -4.28 \\ 4.89 \end{bmatrix}, [Q_{4}] = \begin{bmatrix} 0.88 \\ -1.54 \\ 1.79 \\ -0.95 \end{bmatrix}$ 

# Storey shear forces in each mode

The peak shear force is given by,

$$\mathbf{V}_{ik} = \sum_{j=i+1}^{n} Q_{ik}$$

The storey shear forces for the first mode is,

$$\mathbf{V}_{i1} = \sum_{j=i+1}^{n} Q_{i1} = \begin{bmatrix} V_{11} \\ V_{21} \\ V_{31} \\ V_{41} \end{bmatrix} = \begin{bmatrix} (Q_{11} + Q_{21} + Q_{31} + Q_{41}) \\ (Q_{21} + Q_{31} + Q_{41}) \\ (Q_{31} + Q_{41}) \\ (Q_{41}) \end{bmatrix} = \begin{bmatrix} 239.65 \\ 204.96 \\ 140.65 \\ 56.25 \end{bmatrix} \mathbf{kN}$$

Similarly,

$$\mathbf{V}_{i2} = \begin{bmatrix} 14.11 \\ -3.47 \\ -17.71 \\ -11.67 \end{bmatrix}, \mathbf{V}_{i3} = \begin{bmatrix} 3.178 \\ -4.55 \\ 0.61 \\ 4.89 \end{bmatrix}, \mathbf{V}_{i4} = \begin{bmatrix} 0.18 \\ -0.7 \\ 0.84 \\ -0.95 \end{bmatrix}$$

# Storey shear force due to all modes

Square root of sum of squares (SRSS)  

$$V_{1} = [(V_{11})^{2} + (V_{12})^{2} + (V_{13})^{2} + (V_{14})^{2}]^{1/2}$$

$$= [(239.65)^{2} + (14.11)^{2} + (3.178)^{2} + (0.18)^{2}]^{1/2} = 240.08 \text{kN}$$

$$V_{2} = [(V_{21})^{2} + (V_{22})^{2} + (V_{23})^{2} + (V_{24})^{2}]^{1/2}$$

$$= [(204.96)^{2} + (-3.47)^{2} + (-4.55)^{2} + (-0.7)^{2}]^{1/2} = 205.04 \text{kN}$$

$$V_{3} = [(V_{31})^{2} + (V_{32})^{2} + (V_{33})^{2} + (V_{34})^{2}]^{1/2}$$

$$= [(140.65)^{2} + (-17.71)^{2} + (0.61)^{2} + (0.84)^{2}]^{1/2} = 141.76 \text{kN}$$

$$V_{4} = [(V_{41})^{2} + (V_{42})^{2} + (V_{43})^{2} + (V_{44})^{2}]^{1/2}$$

$$= [(56.25)^{2} + (-11.67)^{2} + (4.89)^{2} + (-0.95)^{2}]^{1/2} = 57.66 \text{kN}$$

#### Lateral forces at each storey due to all modes

Square root of sum of squares (SRSS)

 $F_{roof} = F_4 = V_4 = 57.66 kN$ 

 $F_{floor3} = F_3 = V_3 - V_4 = 141.76-57.66 = 84.1 \text{kN}$ 

 $F_{floor2} = F_2 = V_2 \text{ - } V_3 = 205.04\text{--}141.76 = 63.28 \text{kN}$ 

 $F_{floor1} = F_1 = V_1 - V_2 = 240.08\text{--}205.04 = 35.04 \text{kN}$ 

Table 3.2: Comparison of lateral forces by static and dynamic analysis

	By Equivalent Static Lateral	By Response Spectrum Method		
Force Method (kN) By not		By not considering the	By considering the	
		stiffness of infills (kN)	stiffness of infills (kN)	
Roof	79.96	36.69	57.66	
3 <sup>rd</sup> Floor	73.75	47.26	84.1	
2 <sup>nd</sup> Floor	32.78	33.66	63.28	
1 <sup>st</sup> Floor	8.19	20.93	35.04	

From the above results it was conclude that the value of forces that building can take are coming less in dynamic analysis as compared to static analysis. This shows that if we design building using dynamic analysis though we have to perform more calculations but it give more precise results as compared. So we can say that dynamic analysis is economic

# Chapter 4 Seismic Analysis by Staad Pro

#### 4.1 About Staad Pro

STAAD or (STAAD.Pro) is a structural analysis and design computer program originally developed by Research Engineers International at Yorba Linda, CA in year 1997. In late 2005, Research Engineers International was bought by Bentley Systems. An older version called Staad-III for windows is used by Iowa State University for educational purposes for civil and structural engineers. Initially it was used for DOS-Window system. The commercial version STAAD.Pro is one of the most widely used structural analysis and design software. It supports several steel, concrete and timber design codes. It can make use of various forms of analysis from the traditional 1st order static analysis, 2nd order p-delta analysis, geometric nonlinear analysis or a buckling analysis. It can also make use of various forms of dynamic analysis from modal extraction to time history and response spectrum analysis. In recent years it has become part of integrated structural analysis and design solutions mainly using an exposed API called OpenSTAAD to access and drive the program using an VB macro system included in the application or other by including OpenSTAAD functionality in applications that themselves include suitable programmable macro systems. Additionally STAAD.Pro has added direct links to applications such as RAM Connection and STAAD.Foundation to provide engineers working with those applications which handle design post processing not handled by STAAD.Pro itself. Another form of integration supported by STAAD.Pro is the analysis schema of the CIMsteel Integration Standard, version 2 commonly known as CIS/2 and used by a number modelling and analysis applications.

#### 4.2 Static Analysis

The above analysis done manually is done by staad pro below and results are obtained.

#### 4.2.1 Source code

**Staad Space Start Job Information** Engineer Date 04-Apr-16 **End Job Information** Input Width 79 Unit Meter Kn Joint Coordinates 1 0 0 0; 2 5 0 0; 3 10 0 0; 4 0 3.5 0; 5 5 3.5 0; 6 10 3.5 0; 7 0 7 0; 8 5 7 0; 9 10 7 0; 10 0 10.5 0; 11 5 10.5 0; 12 10 10.5 0; 13 0 14 0; 14 5 14 0; 15 10 14 0; 16 0 0 2.5; 17 5 0 2.5; 18 10 0 2.5; 19 0 3.5 2.5; 20 5 3.5 2.5; 21 10 3.5 2.5; 22 0 7 2.5; 23 5 7 2.5; 24 10 7 2.5; 25 0 10.5 2.5; 26 5 10.5 2.5; 27 10 10.5 2.5; 28 0 14 2.5; 29 5 14 2.5; 30 10 14 2.5; 31 0 0 5; 32 5 0 5; 33 10 0 5; 34 0 3.5 5; 35 5 3.5 5; 36 10 3.5 5; 37 0 7 5; 38 5 7 5; 39 10 7 5; 40 0 10.5 5; 41 5 10.5 5; 42 10 10.5 5; 43 0 14 5; 44 5 14 5; 45 10 14 5; Member Incidences 1 4 5; 2 5 6; 3 7 8; 4 8 9; 5 10 11; 6 11 12; 7 13 14; 8 14 15; 9 1 4; 10 2 5; 11 3 6; 12 4 7; 13 5 8; 14 6 9; 15 7 10; 16 8 11; 17 9 12; 18 10 13; 19 11 14; 20 12 15; 21 19 20; 22 20 21; 23 22 23; 24 23 24; 25 25 26; 26 26 27; 27 28 29; 28 29 30; 29 16 19; 30 17 20; 31 18 21; 32 19 22; 33 20 23; 34 21 24; 35 22 25; 36 23 26: 37 24 27; 38 25 28; 39 26 29; 40 27 30; 41 34 35; 42 35 36; 43 37 38; 44 38 39; 45 40 41; 46 41 42; 47 43 44; 48 44 45; 49 31 34; 50 32 35; 51 33 36; 52 34 37; 53 35 38; 54 36 39; 55 37 40; 56 38 41; 57 39 42; 58 40 43; 59 41 44; 60 42 45; 61 4 19; 62 5 20; 63 6 21; 64 7 22; 65 8 23; 66 9 24; 67 10 25; 68 11 26; 69 12 27; 70 13 28; 71 14 29; 72 15 30; 73 19 34; 74 20 35; 75 21 36; 76 22 37; 77 23 38; 78 24 39; 79 25 40; 80 26 41; 81 27 42; 82 28 43; 83 29 44; 84 30 45: **Element Incidences Shell** 85 13 14 29 28; 86 28 43 44 29; 87 29 30 45 44; 88 10 11 26 25; 89 11 12 27 26; 90 25 26 41 40; 91 26 27 42 41; 92 7 8 23 22; 93 22 37 38 23; 94 23 24 39 38; 95 8 9 24 23; 96 4 5 20 19; 97 5 6 21 20; 98 19 20 35 34; 99 20 21 36 35; 100 14 15 30 29; **Element Property** 85 To 100 Thicknesses 0.15 **Define Material Start Isotropic Concrete** E 2.17185e+007 Poisson 0.17 Density 23.5616 Alpha 1e-005 Damp 0.05 Type Concrete Strength Fcu 27579 End Define Material Member Property American 9 To 20 29 To 40 49 To 60 Pris Yd 0.5 Zd 0.3 61 To 84 Pris Yd 0.45 Zd 0.3 1 To 8 21 To 28 41 To 48 Pris Yd 0.4 Zd 0.3 Constants Material Concrete All **Supports** 1 To 3 16 To 18 31 To 33 Fixed Define 1893 Load Zone 0.36 Rf 5 I 1 Ss 1 St 1 Selfweight 1 Floor Weight Yrange 0 0 Fload 1.75 Xrange 3.5 10 Zrange 0 0 Load 1 Loadtype Seismic Title El X +Ve 1893 Load X 1 Load 2 Loadtype Seismic Title El X -Ve 1893 Load X -1 Load 3 Loadtype Seismic Title El Z+Ve 1893 Load Z 1 Load 4 Loadtype Seismic Title El Z-Ve 1893 Load Z -1 Load 5 Loadtype Dead Title Dl Selfweight Y -1 Member Load 1 To 8 21 To 28 41 To 48 61 To 84 Uni Gy -56 Load 6 Loadtype Live Title Ll Floor Load Yrange 3.5 10.5 Fload -1.75 Gy Load Comb 7 1.5(Dl + Ll + El X + Ve)1 1.5 5 1.5 6 1.5 Load Comb 8 1.5(Dl + Ll + El Z + Ve)

3 1.5 5 1.5 6 1.5 Load Comb 9 1.2(Dl + Ll -El X-Ve) 5 1.2 6 1.2 2 -1.2 Load Comb 10 1.2(Dl + Ll -El Z-Ve) 5 1.2 6 1.2 4 -1.2 Perform Analysis Perform Analysis Print Statics Check Load List 7 To 10 Print Joint Displacements List All Print Member Forces List All Print Support Reaction List 1 To 3 16 To 18 31 To 33 Print Story Drift Print Analysis Results Finish

#### 4.2.2 Results

4.2.2.1

#### Table 4.1: Beam member forces

Beam no.	Max. moment in Z direction (M <sub>z</sub> ) (kNm)	Max. shear force (kN)
1	215.2	239.92
2	198.23	232.11
3	208.33	236.89
4	207.71	235.74
5	196.88	231.85
6	209.1	236.12
7	192.8	229.21
8	184.43	225.53
21	217.87	244.79
22	201.99	237.28
23	210.68	241.59
24	212.18	241.14
25	199.96	236.87
26	213.94	241.71
27	189.03	227.46
28	182.26	224.42
41	215.2	239.92
42	198.23	232.11
43	208.33	236.89
44	207.71	235.74
45	196.88	231.85
46	209.1	236.12
47	192.8	229.21
48	184.43	225.63
61	252.33	141.5
62	242.73	137.78
63	252.33	141.5
64	233.64	132.34
65	222.78	128
66	233.64	132.34
67	223.605	123.37

68	228.36	127.82
69	221.85	123.34
70	205.335	115.86
71	205.14	117.08
72	204.33	116.14
73	264.105	137.92
74	255.345	135.27
75	264.105	137.92
76	270.69	143.7
77	265.14	142.53
78	270.69	143.7
79	263.22	141.83
80	261.765	142.66
81	263.22	141.83
82	221.775	123.54
83	218.775	122.99
84	221.75	123.54

#### 4.2.2.2

#### Table 4.2: Column member forces

Column no.	Max. moment in Z direction (M <sub>z</sub> ) (kNm)	Max. moment in Y direction (M <sub>y</sub> ) (kNm)	Max. shear force (P <sub>u</sub> ) (kN)
9	65.72	27.71	1461.79
10	47.53	28.18	2528.01
11	83.82	27.71	1551.8
12	93.2	19.04	1109.05
13	42.51	21.57	1897.6
14	120.4	21.74	1171.93
15	85.38	22.19	741.4
16	37.21	24.27	1263.47
17	107.21	24.69	775.48
18	148.25	29.22	359.68
19	27.07	32.12	628.44
20	165.48	31.42	371.26
29	68.91	37.76	1960.17
30	50.32	38.4	3033.68
31	87.99	37.76	2022.94
32	97.35	38.75	1453.15
33	51.02	40.57	2231.64
34	126.12	38.75	1495.34
35	87.59	32.64	959.09
36	44.04	34.02	1474.06
37	110.88	32.64	980.68
38	143.39	20.36	472.36
39	25.35	20.92	747.39
40	161.61	20.36	479.12
49	57.34	39.36	1610.4
50	47.53	41.02	2641.98
51	83.82	39.36	1610.4
52	101.21	47.18	1208.6
53	42.51	50.69	1972.25

54	120.4	47.18	1208.6
55	91.77	47.05	793.17
56	37.21	49.66	1301.25
57	107.21	47.05	793.17
58	152.16	45.41	375.5
59	27.07	48.13	639.19
60	165.48	45.41	375.5

### 4.3 Response Spectrum Analysis

The above analysis done manually is done by staad pro below and results are obtained.

#### 4.3.1 Source code

Density 23.5616 Alpha 1e-005 Damp 0.05 **Type Concrete** Strength Fcu 27579 **End Define Material** Member Property American 9 To 20 29 To 40 49 To 60 Pris Yd 0.5 Zd 0.3 61 To 84 Pris Yd 0.45 Zd 0.3 1 To 8 21 To 28 41 To 48 Pris Yd 0.4 Zd 0.3 Constants Material Concrete All **Supports** 1 To 3 16 To 18 31 To 33 Fixed Define 1893 Load Zone 0.36 Rf 5 I 1 Ss 1 St 1 Selfweight 1 Floor Weight Yrange 3.5 10.5 Fload 1.75 Load 1 Loadtype Live Title Load Case 1 Live Floor Load Yrange 3.5 10.5 Fload -1.75 Gy Load 2 Loadtype Seismic Title Load Case 3 Rs Self-weight X 1 Selfweight Y 1 Selfweight Z 1 Member Load 1 To 84 Uni Gx 56 1 To 84 Uni Gz 56 Floor Load Yrange 0 0 Fload 1.75 Gx Yrange 0 0 Fload 1.75 Gy Yrange 0 0 Fload 1.75 Gz Spectrum Srss 1893 X 0.036 Acc Damp 0.05 Soil Type 1 Load 3 Loadtype Dead Title Load Case 2 Dead Selfweight Y -1 Member Load 1 To 84 Uni Gy -56 Perform Analysis Perform Analysis Print Mode Shapes Print Analysis Results Finish

# 4.3.2 Results

#### 4.3.2.1

#### Table 4.3: Beam member forces

Beam no.	Max. moment in Z	Max. shear
	direction (M <sub>z</sub> )	force
	(kNm)	( <b>k</b> N)
1	124.7	149.57
2	124.7	149.57
3	119.69	147.42
4	119.69	147.42
5	123.07	148.55
6	123.07	148.55
7	123.3	150.52
8	123.3	150.52
21	124.34	149.39
22	124.34	149.39
23	119.15	147.15
24	119.15	147.15
25	123.79	148.79
26	123.79	148.79
27	122.04	149.85
28	122.04	149.85
41	124.7	149.57
42	124.7	149.57
43	119.69	147.42
44	119.69	147.42
45	123.07	148.55
46	123.07	148.55
47	123.3	150.52
48	123.3	150.52
61	26.87	75.22
62	26.22	74.9
63	26.87	75.22
64	28.55	77.55
65	29.44	78.39
66	28.55	77.55
67	31.53	80.61
68	33.28	82
69	31.53	80.61
70	24.15	77.01
71	23.89	77.32
72	24.15	77.01
73	26.22	75.22
74	26.87	74.9
75	26.87	75.22
76	28.55	77.55
77	29.44	78.39
78	28.55	77.55
79	31.53	80.61
80	33.28	82

81	31.53	80.61
82	24.15	77.01
83	23.89	77.32
84	24.15	77.01

#### 4.3.2.2

#### Table 4.4: Column member forces

Column no.	Max. moment in Z	Max. moment in Y	Max. shear
	direction (M <sub>z</sub> )	direction (M <sub>y</sub> )	force (P <sub>u</sub> )
	(kNm)	(kNm)	(kN)
9	91.05	7.65	1770.69
10	104.96	7.99	2433.1
11	91.05	7.65	1770.69
12	62.67	13.16	1335.94
13	83.71	13.55	1828.45
14	62.67	13.16	1335.94
15	56.97	15.13	891.06
16	69.37	15.23	1221.59
17	56.97	15.13	891.06
18	99.39	19.77	438.92
19	52.18	20.68	614.33
20	99.39	19.77	438.92
29	95.08	37.76	2061.2
30	110.78	38.4	2746.93
31	95.08	37.76	2061.2
32	61.23	38.75	1537.44
33	99.49	40.57	2038.18
34	61.23	38.75	1537.44
35	54.9	32.64	1022.10
36	81.87	34.02	1357.55
37	54.9	32.64	1022.1
38	94.63	20.36	511.39
39	60.97	20.92	687.46
40	94.63	20.36	511.39
49	44.25	39.36	149.57
50	104.96	41.02	2433.1
51	91.05	39.36	1770.69
52	62.67	47.18	1335.94
53	83.71	50.69	1828.45
54	62.67	47.18	1335.94
55	56.97	47.05	891.06
56	69.37	49.66	1221.59
57	56.97	47.05	891.06
58	99.39	45.41	438.92
59	52.18	48.13	614.33
60	99.39	45.41	438.92

From the above results it was concluded that the member forces of building are coming less in dynamic analysis as compared to static analysis. This shows that if design of building is done using dynamic analysis though to perform more calculations but it give more precise results as compared. So it can be said that dynamic analysis is economic.

# Chapter 5 Design of beams and columns by spreadsheet

# 5.1 Design of beams

Doubly reinforced beams are generally resorted to in situations where the cross-sectional dimensions of the beam are restricted and where singly reinforced sections are not adequate in terms of moment resisting capacity. Doubly reinforced beams are also used in situations where reversal of moments is likely. The presence of compression reinforcement reduces long term deflections due to shrinkage and creep. All compression reinforcement must be enclosed by closed stirrups in order to prevent their possible buckling and to provide some ductility by confinement of concrete.

# 5.2 Spreadsheet of design of doubly reinforced beam

# List of symbols

- b = breadth of beam (mm)D = depth of beam (mm)d' = effective depth of beam (mm) $M_{u} = moment (kNm)$  $A_{st}$  = area of steel in tension (mm<sup>2</sup>)  $n_t$  = number of bars in tension  $Ø_t$  = diameter of bars in tension (mm)  $A_{sc}$  = area of steel in compression (mm<sup>2</sup>)  $n_c$  = number of bars in compression  $Ø_c$  = diameter of bars in compression (mm)  $Ø_{\tau}$  = diameter of stirrups (mm)  $n_{sl}$  = number of ties S = spacing of stirrups (mm/cc) $D_x$  = breadth of column (mm)  $D_v = depth of column (mm)$  $M_{ux}$  = moment in X direction (kNm)
- $M_{uv}$  = moment in Y direction (kNm)
- $P_u$  = axial load on compression member (kN)

Table 5.1: I	Design of b	eam
	value	units
breadth of beam (b)	300	mm
depth of beam (D)	400	mm
characteristic compressive strength of		
concrete(f <sub>ck</sub> )	25	MPa
characteristic strength of steel(fy)	415	MPa
effective depth (d)	350	mm
effective span(l)	6000	mm
factored moment(M <sub>u</sub> )	215.2	kNm
$\mathbf{M}_{\mathbf{u},\lim}$	127.62	kNm
	Mu >	
check for singly or double reinforced beam	Mu,lim	Doubly reinforced
x <sub>u,max</sub> /d	0.48	
Pt,lim	1.2	%
	27	

Determining A			
Determining A <sub>st</sub>			
balanced section where $x_u = x_{u,max}$		2	
A <sub>st,lim</sub>	1260	mm <sup>2</sup>	
diameter of bars for tension steel(Ø)	20	mm	
diameter of stirrups( $\emptyset_{\tau}$ )	6	mm	
clear cover provided to the reinforcement	30	mm	
effective cover to reinforcement(d')	46	mm	
$(\Delta A_{st})_{reqd}$	798	mm <sup>2</sup>	
(A <sub>st</sub> ) <sub>reqd</sub>	2058	mm <sup>2</sup>	
number of bars	3		
Øread	30	mm	
A <sub>ct</sub>	2121	mm <sup>2</sup>	
actual d	349	mm	
as actual d is less than the d assumed earlier	r so revising	g the above calculations by	
d=349mm		•	
$\mathbf{M}_{\mathbf{u},\mathbf{lim}}$	126.89	kNm	
A <sub>st.lim</sub>	1256.4	mm <sup>2</sup>	
$(\Delta A_{st})_{reqd}$	801	mm <sup>2</sup>	
(A <sub>st</sub> ) <sub>regd</sub>	2057.4	mm <sup>2</sup>	
Actual $(\Delta A_{st})_{\text{provided}}$	864.6	mm <sup>2</sup>	
Determining A <sub>sc</sub>			
assuming x <sub>u</sub> =x <sub>u</sub> max			
d'/d	0.132		
from table A			
fsc	345.82	MPa	
(Acc)road	933	mm <sup>2</sup>	
number of bars	5		
Ørrad	16	mm	
A	1006	mm <sup>2</sup>	
ra <sub>sc</sub>	1000		
Design Check			
assuming $x_n < x_n$ it suffices to establish $n_n > \infty$			
$\mathbf{p}_c^*$			
actual d provided	349	mm	
d'	44	mm	
d'/d	0.127		
from table A			
f <sub>sc</sub>	346.77	MPa	
actual p <sub>t</sub>	2.026	%	
actual p <sub>c</sub>	0.961	%	
	20	1	
	38		

pc*	0.889	%		
as $p_c > p_c^*$ , Hence OK				
Check for deflection control				
<b>p</b> <sub>t</sub>	2.026	%		
f <sub>s</sub>	234	MPa		
kt (from fig. 4 of code IS 456:2000)	0.8468			
Pc	0.961	%	<u> </u>	
k <sub>c</sub> (from fig.5 of code IS 456:2000)	1.2			
(l/d) <sub>max</sub>	20.33			
(I/d) <sub>provided</sub>	17.2			
as $(l/d)_{max} > (l/d)_{provided}$ , Hence OK				
Shear Design				
max. shear stress ( $\tau_{cmax}$ )	3.1	MPa		
factored shear force (V <sub>u</sub> )	230.37	kN	「 <u> </u>	
nominal shear stress ( $ au_v$ )	2.21	MPa		
β	1.44			
design shear strength of concrete $(\tau_c)$	0.82	MPa		
check whether $\tau_{c,max} > \tau_c$	Yes			
shear force resisted by concrete $(V_{uc})$	85.86	kN		
		Provide 'calculated shear		
check whether $V_u > V_{uc}$	Yes	rft.'	<b> </b>	
diameter of stirrups( $\emptyset_{\tau}$ )	6	mm		
number of stirrup legs	2			
Minimum shear reinforcement				
stirrup spacing along beam length	171	mm/cc		
check whether stirrup spacing < 300mm	Yes		[	
			171	mm
provide stirrups of diameter	6 mm	2 legged	c/c	
Calculated shear rainforcement				
$chean fance taken un by stimming (V_{i})$	144 51	1-N		
shear force taken up by surrups ( $v_{us}$ ) stirrun spacing along heam length	50	mm/cc		
check whether stirrup spacing < 300mm	Yes			
			50	mm
provide stirrups of diameter	6 mm	2 legged	c/c	

# 5.3 Design of beams of building

# 5.3.1

# Table 5.2: Beams in transverse direction

Beam No.	bxD	d	Mu	M <sub>u,lim</sub>	A <sub>st</sub>	n <sub>t</sub>	Øt	A <sub>sc</sub>	n <sub>c</sub>	Øc	Shear reinfo	orcem	ent
	mm	mm	kNm	kNm	mm <sup>2</sup>		mm	mm <sup>2</sup>		mm			
											Øτ	n <sub>sl</sub>	S
											mm		mm/ cc
1	300x4 00	349	215.2	127.62	2121	3	30	1006	5	16	6	2	50
2	300x4 00	349. 5	198.2 3	127.62	1982	3	29	884	5	15	6	2	49
3	300x4 00	349	208.3 3	127.62	2121	3	30	1006	5	16	6	2	48
4	300x4 00	349	207.7 1	127.62	2121	3	30	1006	5	16	6	2	48
5	300x4 00	349. 5	196.8 8	127.62	1982	3	29	884	5	15	6	2	49
6	300x4 00	349	209.1	127.62	2121	3	30	1006	5	16	6	2	48
7	300x4 00	349. 5	192.8	127.62	1982	3	29	884	5	15	6	2	50
8	300x4 00	350	184.4 3	127.62	1848	3	28	664	5	13	6	2	50
21	300x4 00	349	217.8 7	127.62	2121	3	30	1006	5	16	6	2	45
22	300x4 00	349. 5	201.9 9	127.62	1982	3	29	884	5	15	6	2	47
23	300x4 00	349	210.6 8	127.62	2121	3	30	1006	5	16	6	2	46
24	300x4 00	349	212.1 8	127.62	2121	3	30	1006	5	16	6	2	46
25	300x4 00	349. 5	199.9 6	127.62	1982	3	29	884	5	15	6	2	47
26	300x4 00	349	213.9 4	127.62	2121	3	30	1006	5	16	6	2	46
27	300x4 00	350	189.0 3	127.62	1848	3	28	664	5	13	6	2	50
28	300x4 00	350	182.2 6	127.62	1848	3	28	664	5	13	6	2	51
41	300x4 00	349	215.2	127.62	2121	3	30	1006	5	16	6	2	47
42	300x4 00	349. 5	198.2 3	127.62	1982	3	29	884	5	15	6	2	49
43	300x4 00	349	208.3 3	127.62	2121	3	30	1006	5	16	6	2	48
						40							

44	300x4 00	349	207.7 1	127.62	2121	3	30	1006	5	16	6	2	48
45	300x4 00	349. 5	196.8 8	127.62	1982	3	29	884	5	15	6	2	49
46	300x4 00	349	209.1	127.62	2121	3	30	1006	5	16	6	2	48
47	300x4 00	349. 5	192.8	127.62	1982	3	29	884	5	15	6	2	50
48	300x4 00	350	184.4 3	127.62	1848	3	28	664	5	13	6	2	50

5.3.2

# Table 5.3: Beams in longitudinal direction

Beam No	bxD	d	Mu	M <sub>u,lim</sub>	A <sub>st</sub>	n <sub>t</sub>	Øt	A <sub>sc</sub>	n <sub>c</sub>	Øc	Shear	r orcem	ent
110.	mm	mm	kNm	kNm	mm <sup>2</sup>		mm	mm <sup>2</sup>		mm	Tenni	orcem	ent
											Øτ	n <sub>sl</sub>	S
											mm		mm/ cc
61	300x4 50	399	252.3 3	166.68	2121	3	30	764	3	18	6	2	56
62	300x4 50	399	242.7 3	166.68	2121	3	30	764	3	18	6	2	57
63	300x4 50	399	252.3 3	166.68	2121	3	30	764	3	18	6	2	56
64	300x4 50	399. 5	233.6 4	166.68	1982	3	29	604	3	16	6	2	59
65	300x4 50	399. 5	222.7 8	166.68	1982	3	29	604	3	16	6	2	60
66	300x4 50	399. 5	233.6 4	166.68	1982	3	29	604	3	16	6	2	59
67	300x4 50	399. 5	223.6 05	166.68	1982	3	29	604	3	16	6	2	63
68	300x4 50	399. 5	228.3 6	166.68	1982	3	29	604	3	16	6	2	61
69	300x4 50	399. 5	221.8 5	166.68	1982	3	29	604	3	16	6	2	63
70	300x4 50	400	205.3 35	166.68	1848	3	28	462	3	14	6	2	65
71	300x4 50	400	205.1 4	166.68	1848	3	28	462	3	14	6	2	65
72	300x4 50	400	204.3 3	166.68	1848	3	28	462	3	14	6	2	65
73	300x4 50	398. 5	264.1 05	166.68	2265	3	31	943	3	20	6	2	58
74	300x4 50	398. 5	255.3 45	166.68	2265	3	31	943	3	20	6	2	59
	50	5	45			41							

75	300x4	398.	264.1	166.68	2265	3	31	943	3	20	6	2	58
	50	5	05										
76	300x4	398.	270.6	166.68	2265	3	31	943	3	20	6	2	56
	50	5	9										
77	300x4	398.	265.1	166.68	2265	3	31	943	3	20	6	2	56
	50	5	4										
78	300x4	398.	270.6	166.68	2265	3	31	943	3	20	6	2	56
	50	5	9										
79	300x4	398.	263.2	166.68	2265	3	31	943	3	20	6	2	56
	50	5	2										
80	300x4	398.	261.7	166.68	2265	3	31	943	3	20	6	2	56
	50	5	65										
81	300x4	398.	263.2	166.68	2265	3	31	943	3	20	6	2	56
	50	5	2										
82	300x4	399.	221.7	166.68	1982	3	29	604	3	16	6	2	63
	50	5	75										
02	200 4	200	010.7	166.60	1000	2	20	<u>(0</u> 4	2	16	6	-	(2)
83	300x4	399. 5	218.7	100.08	1982	3	29	604	3	16	6	2	63
	50	3	15										
84	300x4	399	221.7	166 68	1982	3	29	604	3	16	6	2	63
0-	50074	5	5	100.00	1702			007	5	10		-	05
	50	5	5										
	1	1	1	1	1	1	1	1	1	1	1	1	

# 5.4 Design of columns

A compression member is a structural element which is subjected to axial compressive forces. Compression members are most commonly encountered in reinforced concrete buildings as columns forming part of vertical framing system. Whether the structure is man made or created by nature the key element in resisting collapse under gravity load is the column. The column is representative of all types of compression members and hence sometimes the terms column and compression member are used interchangeably.

# 5.5 Spreadsheet of design of short columns under axial compression with biaxial bending

Table 5.4 Design of	Column	
	values	units
breadth of column (b) (D <sub>x</sub> )	300	mm
depth of column (D) (D <sub>y</sub> )	500	mm
unsupported length of column (l)	3500	mm
Pu	1461.79	kN
M <sub>ux</sub>	65.72	kNm
M <sub>uy</sub>	27.71	kNm
characteristic strength of steel(fy)	415	MPa
characteristic compressive strength of concrete(f <sub>ck</sub> )	25	MPa
Slenderness ratio		

effective length ratio	0.85	
effective length about x-x axis (l <sub>ex</sub> )	2975	mm
effective length about y-y axis (l <sub>ey</sub> )	2975	mm
l <sub>ex</sub> /D <sub>x</sub>	9.92	
$l_{ev}/D_y$	5.95	
check for l <sub>ev</sub> /D <sub>y</sub>	Yes short column	
check for l <sub>ex</sub> /D <sub>x</sub>	Yes short column	
Check minimum eccentricities		
Applied eccentricities		
ex	44.96	mm
e <sub>y</sub>	18.96	mm
Minimum eccentricities		
e <sub>x,min</sub>	20	mm
e <sub>y,min</sub>	23.67	mm
Longitudinal reinforcement		
M <sub>u</sub>	83	kNm
depth of compression reinforcement (d')	50	mm
d'/D	0.1	
Pu	0.39	
m <sub>u</sub>	0.03	
from chart 44 of SP16, p/f <sub>ck</sub>	0.02	
Pregd	0.5	
A <sub>s,reqd</sub>	750	mm <sup>2</sup>
number of bars	6	
Ø	18	mm
A <sub>s</sub>	1527	mm <sup>2</sup>
provide 6 bars of 18mm dia		
Pprovided	1.02	%
p/f <sub>ck</sub>	0.05	
along x-x axis		
d'/D	0.17	
from SP16 value of m <sub>ux</sub>	0.06	
M <sub>ux1</sub>	112.5	kNm
along y-y axis		

from SP16 value of m <sub>uy</sub>	0.08	
M <sub>uy1</sub>	150	kNm
	2145.6	1.53
P <sub>uz</sub>	2145.6	KIN
P <sub>u</sub> /P <sub>uz</sub>	0.69	
α <sub>n</sub>	1.82	
check safety under biaxial loading Transverse reinforcement	0.43	safe
tie diameter (Ø <sub>t</sub> )	6	mm
tie spacing (s <sub>t</sub> )	288	mm
provide 6 Ø ties @ 288c/c		

5.6

# Table 5.5: Design of columns of building

Column No.	$\mathbf{D}_{\mathbf{x}} \mathbf{x}$ $\mathbf{D}_{\mathbf{v}}$	l mm	P <sub>u</sub> kN	M <sub>ux</sub> kNm	M <sub>uy</sub> kNm	M <sub>u</sub> kNm	Longitudinal reinforcement			Transverse reinforcement		
	mm						$A_s mm^2$	Ø m m	No.of bars	Ø <sub>t</sub> mm	S mm/cc	
9	300 x 500	3500	1461.79	65.72	27.71	83	1527	18	6	6	288	
10	300 x 500	3500	2528.01	47.53	28.18	64	3927	25	8	8	300	
11	300 x 500	3500	1551.8	83.82	27.71	102	1527	18	6	6	288	
12	300 x 500	3500	1109.05	93.2	19.04	110	1527	18	6	6	288	
13	300 x 500	3500	1897.6	42.51	21.57	55	1885	20	6	6	300	
14	300 x 500	3500	1171.93	120.4	21.74	141	1527	18	6	6	288	
15	300 x 500	3500	741.4	85.38	22.19	102	1527	18	6	6	288	
					44							

16	300	3500	1263.47	37.21	24.27	52	1527	18	6	6	288
	х										
	500										
17	300	3500	775.48	107.21	24.69	127	1527	18	6	6	288
	X				,				-	-	
	500										
18	300	3500	359.68	148.25	29.22	174	1527	18	6	6	288
10	x	0000	227.00	1.0120			1021	10	Ũ	0	200
	500										
19	300	3500	628 44	27.07	32.12	49	1527	18	6	6	288
17	x	5500	020.11	27.07	52.12		1521	10	Ũ	0	200
	500										
20	300	3500	371.26	165.48	31.42	194	1527	18	6	6	288
20	500 v	3300	571.20	105.40	51.72	174	1527	10	0	0	200
	500										
20	300	3500	1060 17	68.01	27.76	01	2046	25	6	6	200
29	300 v	3300	1900.17	08.91	57.70	91	2940	23	0	0	300
	x 500										
20	200	2500	2022.69	50.22	29.4	72	4000	25	10	6.25	200
50	300	5500	3033.08	50.52	30.4	15	4909	23	10	0.23	300
	X 500										
	300										
21	200	2500	2022.04	97.00	2776	111	2016	25	C	6.25	200
51	300	3500	2022.94	87.99	37.70	111	2946	25	6	6.25	300
	X										
22	500	2500	1450.15	07.05	20.75	101	1507	10	6	6	200
32	300	3500	1453.15	97.35	38.75	121	1527	18	6	6	288
	X										
	500				40.55				-		
33	300	3500	2231.64	51.02	40.57	75	2946	25	6	6.25	300
	X										
	500							1.0	_	_	
34	300	3500	1495.34	126.12	38.75	152	1527	18	6	6	288
	X										
	500										
35	300	3500	959.09	87.59	32.64	108	1527	18	6	6	288
	Х										
	500										
36	300	3500	1474.06	44.04	34.02	64	1527	18	6	6	288
	Х										
	500										
37	300	3500	980.68	110.88	32.64	133	1527	18	6	6	288
	Х										
	500										
38	300	3500	472.36	143.39	20.36	167	1527	18	6	6	288
	Х										
	500										
39	300	3500	747.39	25.35	20.92	38	1527	18	6	6	288
	Х										
	500										
40	300	3500	479.12	161.61	20.36	187	1527	18	6	6	288
	Х										
	500										
					45						

49	300	3500	1610.4	57.34	39.36	80	1527	18	6	6	288
	х										
	500										
50	300	3500	2641.98	47.53	41.02	73	7069	30	10	7.5	300
	Х										
	500										
51	300	3500	1610.4	83.82	39.36	107	1527	18	6	6	288
	X										
52	300	3500	1208.6	101 21	17 18	120	1527	18	6	6	288
52	300 x	3300	1200.0	101.21	47.10	129	1327	10	0	0	200
	500										
53	300	3500	1972.25	42.51	50.69	77	2946	25	6	6.25	300
	Х										
	500										
<b>E</b> 4	200	2500	1000 (	120.4	47 10	1.40	1507	10	6	<i>c</i>	200
54	300 v	3500	1208.6	120.4	47.18	149	1527	18	6	6	288
	x 500										
55	300	3500	793.17	91.77	47.05	119	1527	18	6	6	288
	Х										
	500										
57	200	2500	1201.25	27.01	10.00	70	1507	10	6	~	200
56	300 v	3500	1301.25	37.21	49.66	12	1527	18	6	6	288
	x 500										
57	300	3500	793.17	107.21	47.05	135	1527	18	6	6	288
	X									-	
	500										
58	300	3500	375.5	152.16	45.41	183	1527	18	6	6	288
	X										
50	500	2500	(20.10	27.07	40.12	<i>C</i> 1	1507	10	6	6	200
59	300 v	3500	039.19	27.07	48.13	64	1527	18	0	0	288
	500										
60	300	3500	375.5	165.48	45.41	198	1527	18	6	6	288
	X		-			_	-	_			
	500										

# **Conclusions**

From the work presented above the following conclusions can be made :

- 1. From the Table 3.2 it was concluded that the value of forces that building can take are coming less in dynamic analysis as compared to static analysis. This shows that if design of building is done using dynamic analysis though we have to perform more calculations but it give more precise results as compared.
- 2. While carrying out response spectrum analysis consideration of infill wall in stiffness leads to better estimation of earthquake forces.
- 3. Software analysis and design using staad pro saves a lot of time spent in computation of results.
- 4. From the Table 4.1, Table 4.2, Table 4.3 and Table 4.4 it was concluded that response spectrum analysis is more economical.

In work presented above design, spread sheets for doubly reinforced beam and biaxial column were prepared. Their reinforcement detail is calculated from spread sheet and the results were tabulated.

### **References**

- Pankaj Agarwal and Manish Shrikhande, 2009, Earthquake Resistant Design Of Structures 1<sup>st</sup> Edition.
- S Unnikrishna Pillai and Devdas Menon, 2014, *Reinforced Concrete Design 3<sup>rd</sup> edition*.
- IS 1893-1 (2002): Criteria for Earthquake Resistant Design of Structures, Part 1: General Provisions and Buildings [CED 39: Earthquake Engineering].
- IS 456 (2000): Plain and Reinforced Concrete- Code of Practice (Fourth revision).
- Ankesh Sharma and Biswobhanu Bhadra, 2013, Seismic Analysis And Design Of Vertically RC Building Frames.